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Mission Overview

- Mission objectives
 - responsive multi-spectral imaging
 - real-time control
- Organizations
 - bus design and payload by SFL
 - experimental technologies by SPACE-SI
 - secondary X-band transmitter
 - image processing pipeline
 - real-time interactive ACS





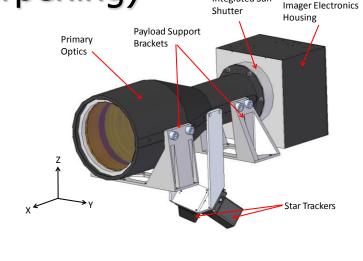
Primary Payload

Primary instrument – high-resolution imaging

- fast 155 mm f/2.3 optics
- 300 lp/mm resolution at focal plane

5 still channels (with pan sharpening)

Channel	Band (nm)	GSD	Swath
PAN	400 – 900	2.8 m	
Blue	420 – 520	5.8 m	
Green	535 – 607	5.8 m	10 km
Red	634 – 686	5.8 m	
Near-IR	750 – 960	5.8 m	



Integrated Sun





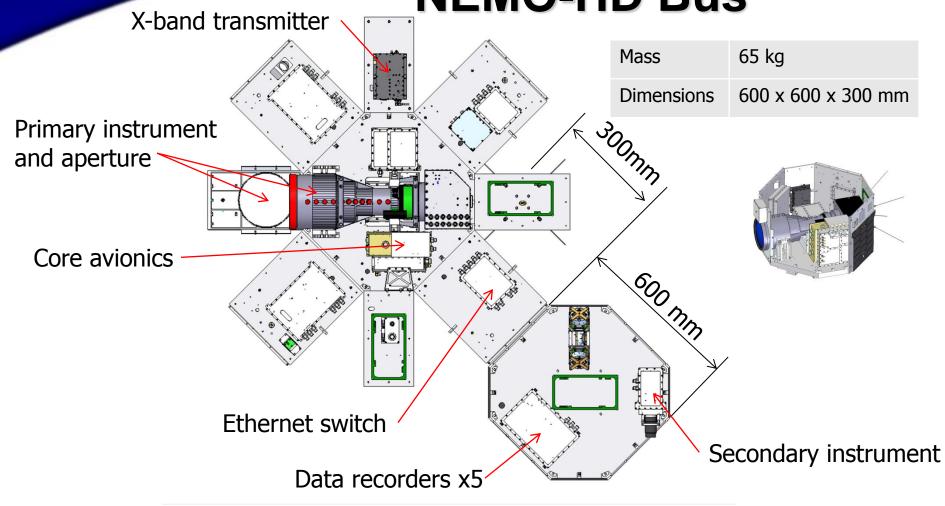
- High-resolution video shares primary optics
- Secondary video for wide-angle context
- Dual real-time or recorded video streams

	Primary	Secondary	
GSD	2.8 m	40 m	
Swath	5.4 km	76.8 km	
Resolution (pixels)	1920 x 1080 (full HD)		
Encoding	H.264		
Data rate	Up to 25 Mb/s per channel		
Chroma	RGB (PAN pass-band)		





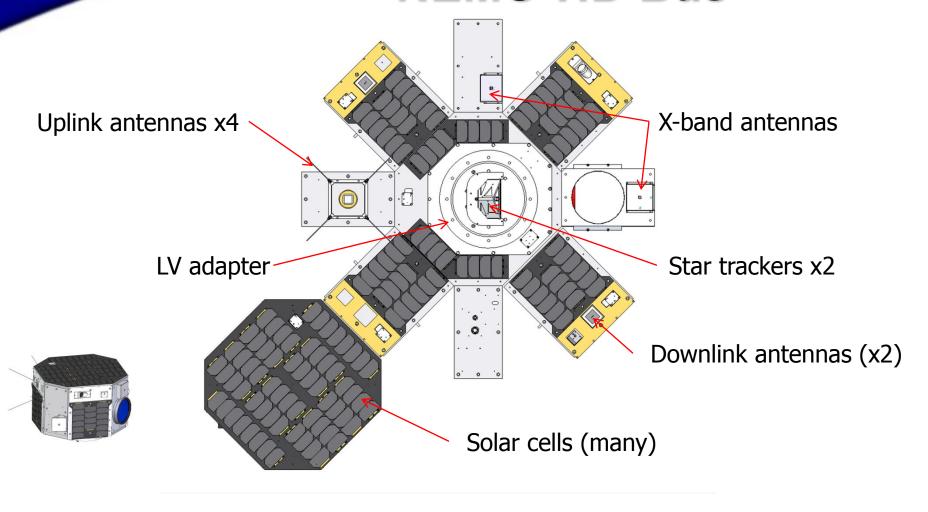








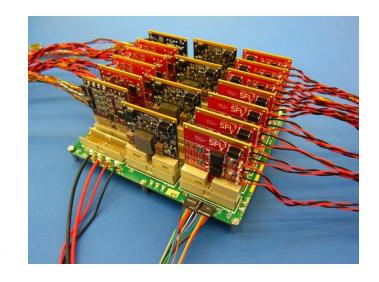
NEMO-HD Bus





NEMO-HD Power, C&DH

- Modular Power System
 - 5 W keep-alive
 - 215 W peak
 - (not shown) 5.8 Ah battery
- Housekeeping and Attitude Computers
 - SFL flight heritage
 - some cross-strapping for redundancy









NEMO-HD ADCS

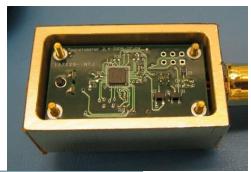
Star trackers



Sun sensors



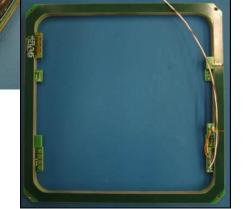
Magnetometer



Reaction wheels

Determination	Pointing	Slew rates
< 15" 1-σ	< 120" 2-σ	1.5 °/s





Magnetic torquer

Wheel, star tracker photos courtesy of Sinclair Interplanetary

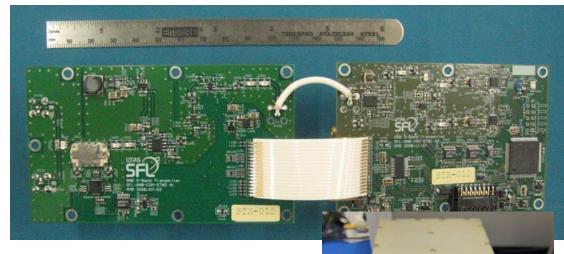


NEMO-HD T&C

UHF receiver



S-band transmitter



Uplink	UHF	4 kb/s
T&C downlink	S-band	up to 1 Mb/s
Payload downlink	X-band	50 Mb/s



Payload Electronics

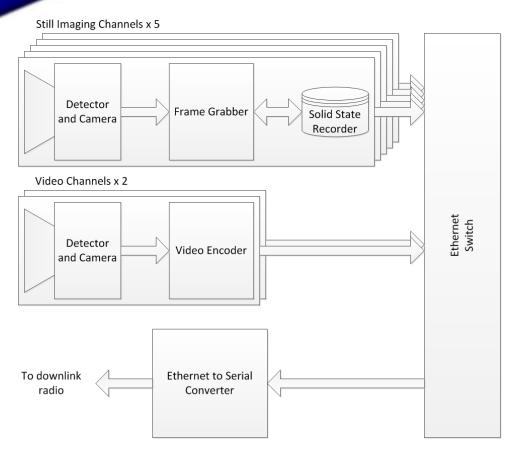
Very short development timeline

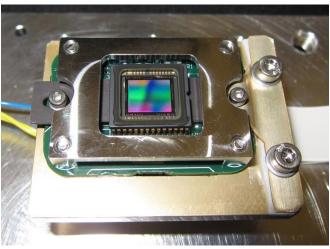
- flight-hardened COTS assemblies
- open-source software
- standard interfaces and protocols
 - Ethernet (100 Mb/s and 1000 Mb/s)
 - Linux
 - H.264 video
 - RTSP
 - GigE Vision
 - SATA





Payload Electronics





Multispectral camera





Payload Electronics



Ethernet-serial converter



Ethernet switch





Action	Real-time	Real-time (auto)	Time-tagged
Select channels	yes	yes	yes
View video	yes	yes	no
Record video	opt	opt	opt
ACS	user	target-based	commanded
Frame/swath acquisition	on command	on command	timed

- Payload data can be downlinked at any point
- Automatic slew to sun point when idle



A Flexible EO Platform

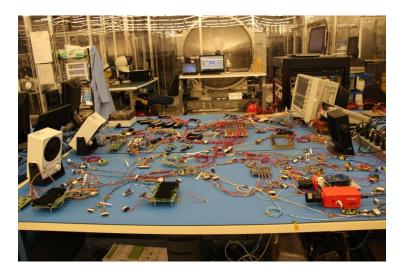
- NEMO-HD bus driven by payload size
- Heavy optics driven by
 - channel count
 - resolution
- Change instrument specs for a smaller spacecraft
- Flexible multi-channel architecture easy to adapt to other missions
- Power systems scales 2 W $< P_{\text{max}} < 1 \text{ kW}$





Current Status

- Structure in manufacturing
- Bus avionics complete and accepted
- Payload electronics mostly accepted
 - will be introduced to optics late 2013
- "Flatsat" integration
- Final assembly 1-2Q2014



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