

# Development of Binary Black Hole Observation Satellite “ORBIS”

---

NISHI, Kentaro

Manager of ORBIS Project  
Space Systems Laboratory, Tokyo Metropolitan University

# Agenda

- Concept of ORBIS
- Mission of ORBIS

## Progress of

- Mission Subsystem
  - Attitude Determination Subsystem
  - Structure Subsystem
  - other Subsystem
- 
- Summary



# Microsatellite for Science Mission

Space Systems Laboratory in Tokyo Metropolitan University is developing a **general-purpose** microsatellite with ease of integration for **science missions**,

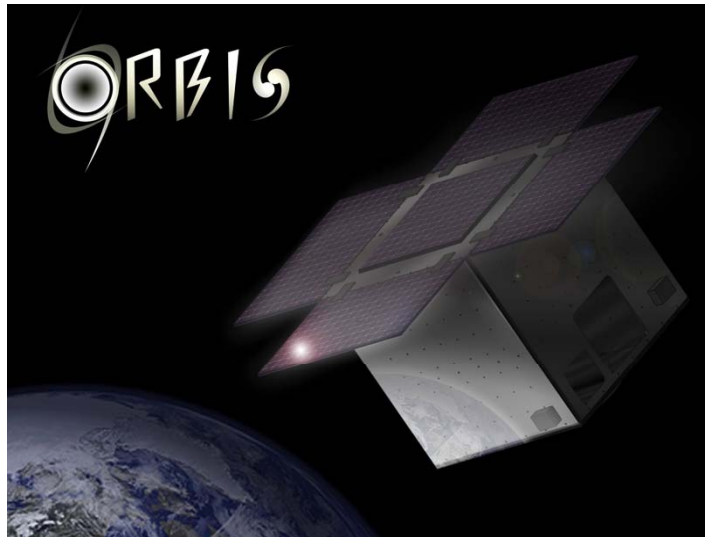
because

- ✓ microsatellites are **the most promising infrastructure** for certain science missions.
- ✓ and the specification for general-purpose application will make satellites **inexpensive and trusted**.

# Microsatellite for Science Mission



We are developing  
ORbiting Binary black hole Investigation Satellite, ORBIS,  
as the first one of the general-purpose microsatellites for  
science mission.



Microsatellite "ORBIS"



# Microsatellite for Science Mission

Microsatellites are **the most promising infrastructure** for certain science missions.

because

✓ there is **no way in the current astronomy** to accomplish such science missions as conduct long-term observations for the specified astral bodies with respect to **budget**, **period**, and **occupancy time**.

✓ and **some scientists expect** an utilization of microsatellite for the observations.

**Investigation of Binary Black Hole** is one of these science mission which is eagerly requiring microsatellite.

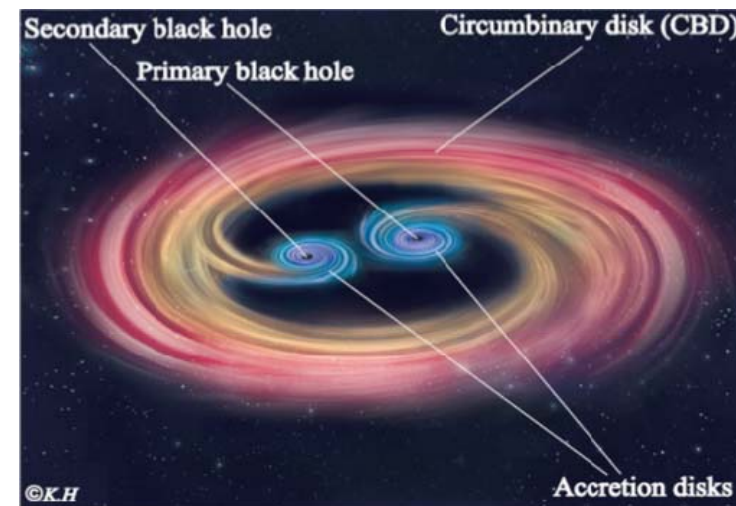
# Mission of ORBIS

Binary Black Hole (BBH) consists of a **pair of black holes** and would contribute to develop kinds of **cosmology**. Past researches are saying that BBH **surely exist** but has **not been observed** yet.

BBH is dark object, and shows **unique cycle variation in X-ray region**.

which requires

- ✓ **long term, continuous**
- ✓ and **high-precision** observation.

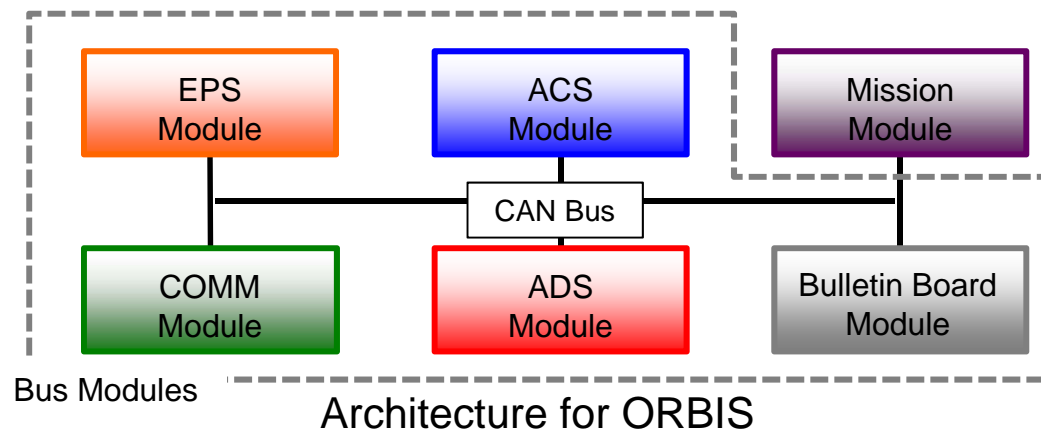


Binary Black Hole

So that microsatellite is and only a way for BBH observation.

# Architecture for General-Purpose

Architecture for ORBIS is based on **distributed autonomous system**, where each subsystem is **modularized into a block** and communicates with the others via **CAN bus**.



Though this architecture is partially realized by **Panel ExTension SATellite** (PETSAT), developed by SOHLA, ORBIS adds another subsystem, **Bulletin Board subsystem**, to make reduction in labor required for system integration and actively utilize **Flag & Play**.

# ORBIS Project



Concept of ORBIS won the first prize at 18<sup>th</sup> Satellite Design Contest in 2010, and we officially started ORBIS Project.

The project team consists of members in TMU, Kogakuin University, Meisei University and ISAS/JAXA.



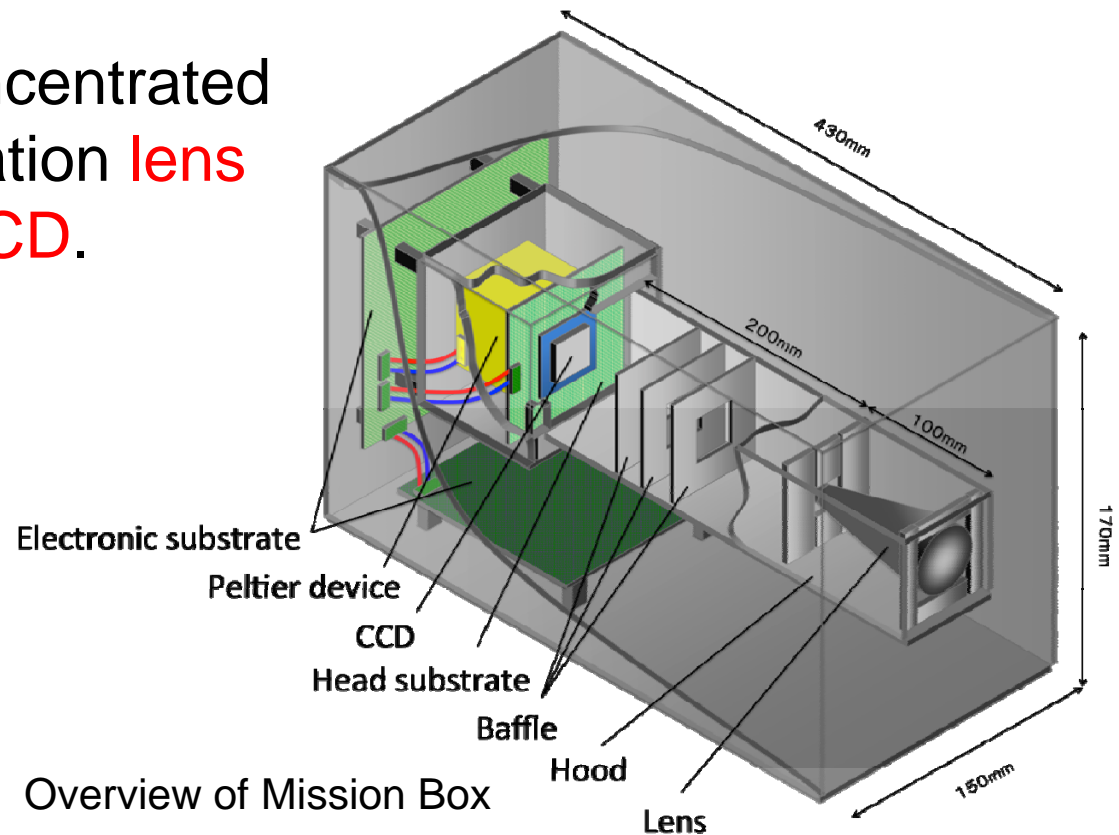
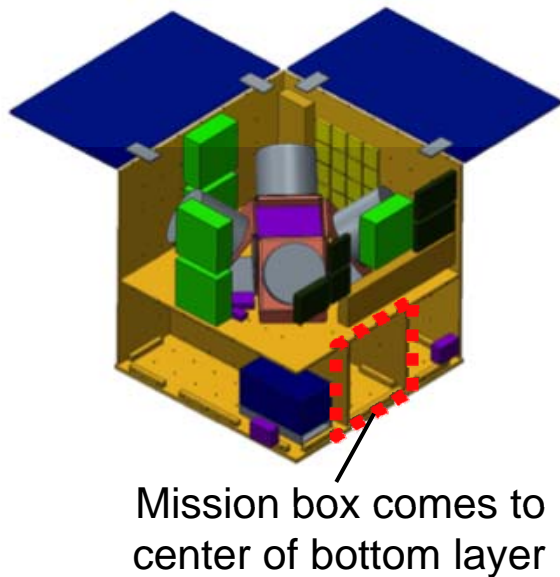
Now, ORBIS is on BBM and Structure Thermal Model phase.



# Progress of Mission Subsystem

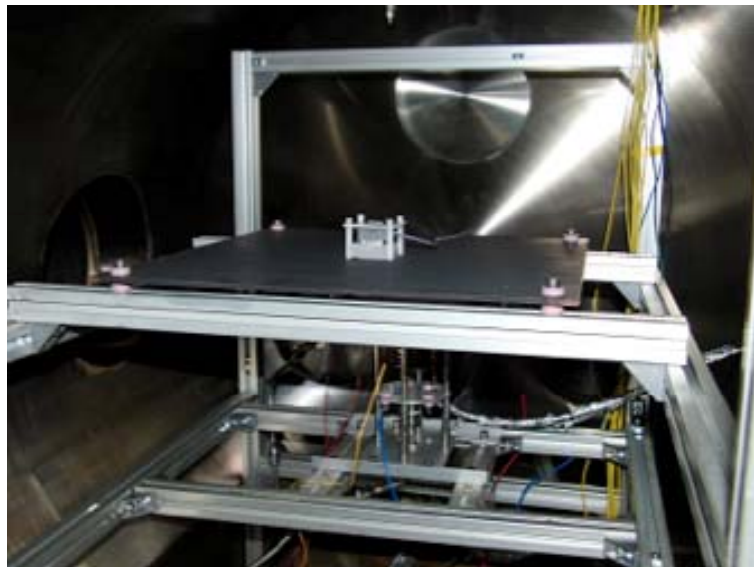
Mission Subsystem with X-ray observation device is designed to be installed into the mission box in ORBIS.

The incident light is concentrated with a kind of concentration lens to be focused on the CCD.

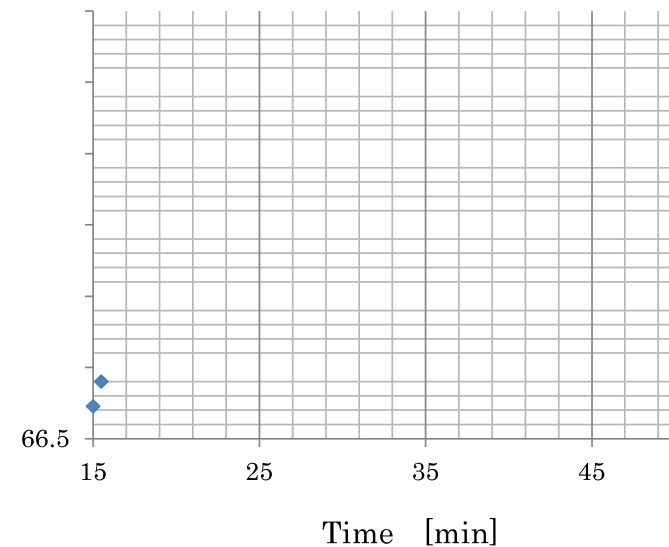


# Progress of Mission Subsystem

- ✓ We conducted some tests on **Peltier device** in a vacuum chamber, and the result CCD can be cooled to  $-26[\text{deg}]$
- ✓ and now we are designing a light-collecting system and circuits for light-receiving system.



Peltier device in a vacuum chamber



Example of result of the test

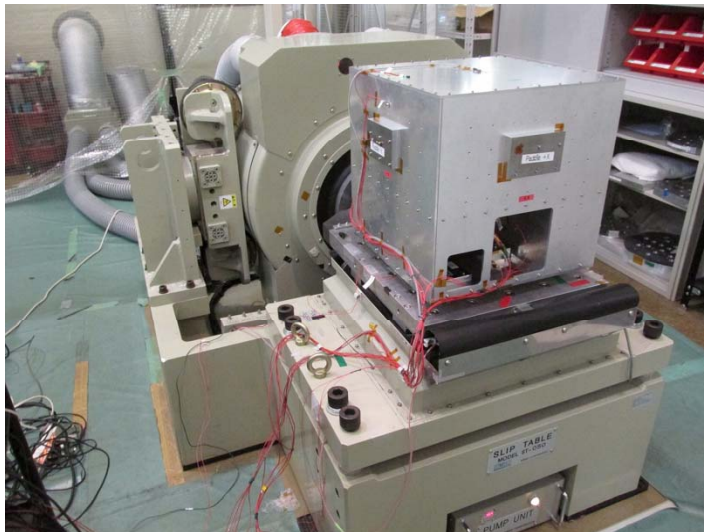
# Progress of ADS

For highly-precise observation, Attitude Determination Subsystem adopts Star Tracker and **Gyroscopes**. In order to judge whether certain gyroscope can be use for ORBIS or not, we conducted a **static experiment** on a gyroscope.

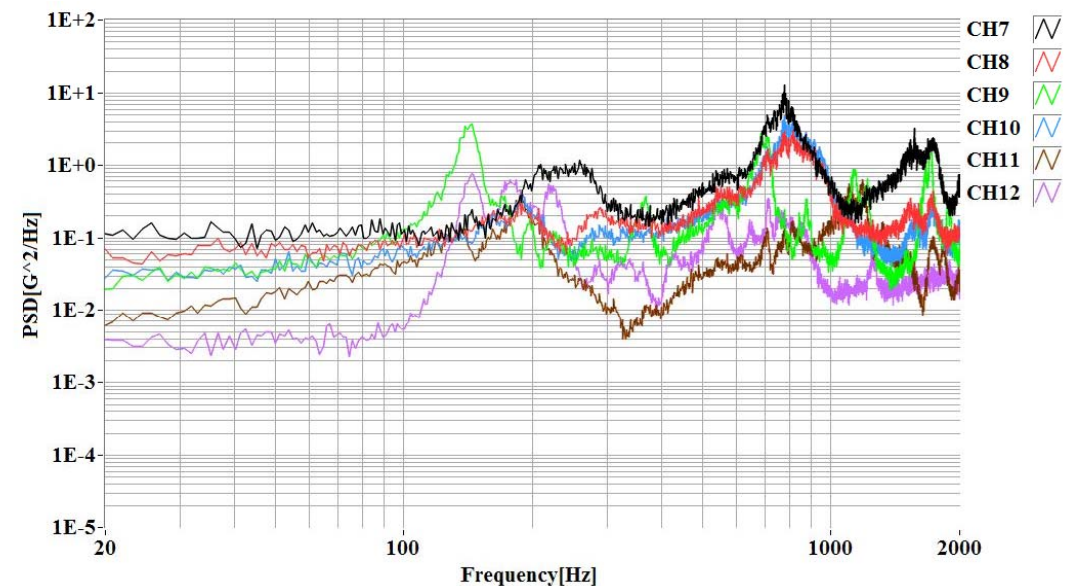
As a result, **Estimate Accuracy of Attitude Angle** is  $9.3 \times 10^{-4}$  [deg] on the gyroscope, showing the gyroscope **can satisfy** ORBIS's requirement, less than  $1.0 \times 10^{-3}$ , at native environment.

# Progress of STR

We designed Structure Thermal Model of ORBIS and conduct series of vibration tests on it under the vibration levels of Qualification Test of H-IIA.



Overview of vibration test  
at Kyushu Institute of Technology

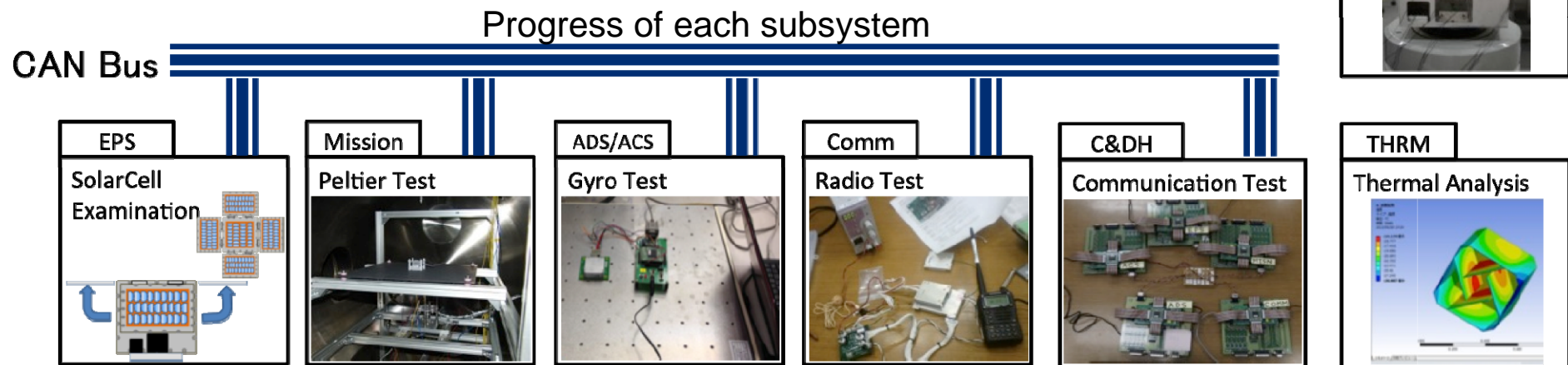


Example of result of vibration test

The result gave us important information to improve the structure.

# Progress of other subsystem

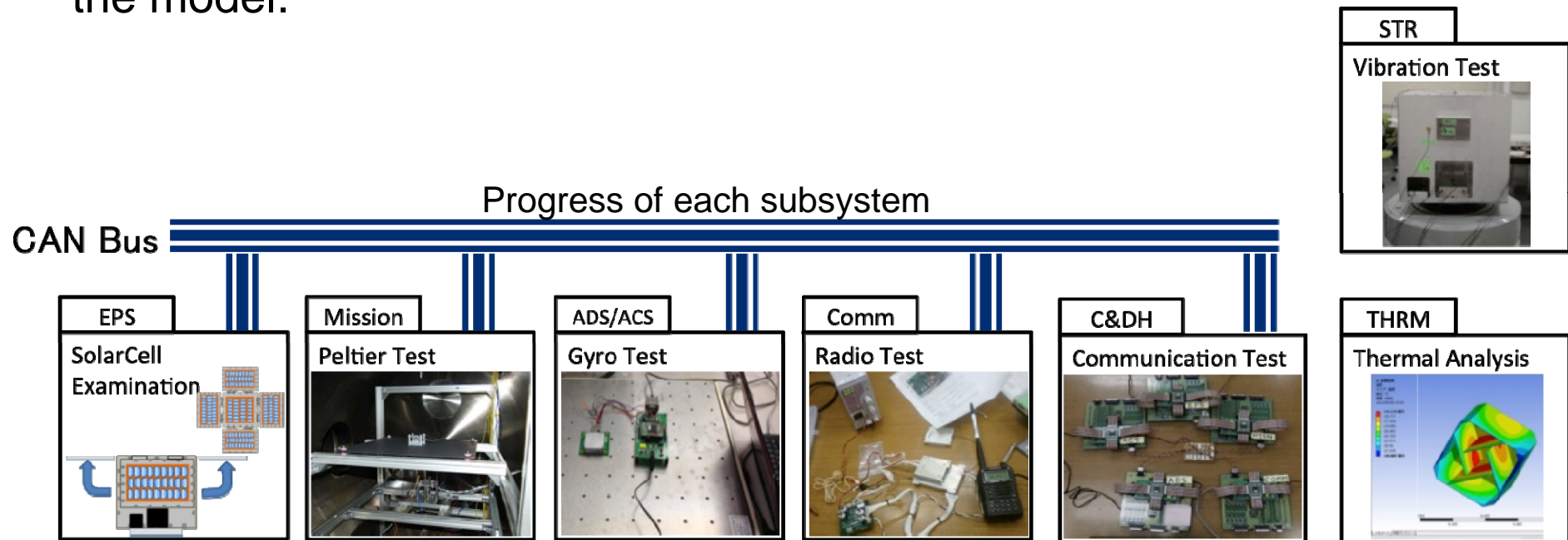
- ✓ C&DH conducted a communication test on CAN bus and now is designing common circuit board of ORBIS which is installed into all the subsystems.
- ✓ EPS estimated the amount of power supply and consumption to distribute power to all the subsystems.
- ✓ ACS analyzed disturbance of reaction wheels and now developing the simulator of entire ADCS.





# Progress of other subsystem

- ✓ COMM analyzed the data traffic for Amateur band and S-band and tested some of them.
- ✓ TCS estimated the amount of thermal input, where the evaluation of contact thermal resistance in the most important and will be tested with the model.



# Summary

- Space Systems Laboratory in Tokyo Metropolitan University is developing a general-purpose microsatellite for science mission, ORBIS.
- We set investigation of the Binary Black Hole as its mission.
- ORBIS is now under BBM phase and each subsystem is developing and testing their own module.

# Development of Binary Black Hole Observation Satellite “ORBIS”

---

NISHI, Kentaro

Manager of ORBIS Project  
Space Systems Laboratory, Tokyo Metropolitan University



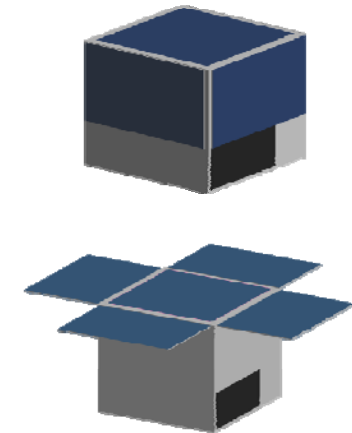
# Specification of ORBIS

## Structure

ORBIS has 4 paddles of EPS.

Its dimension is  $460 \times 460 \times 440$ [mm] in closing  
and  $980 \times 980 \times 440$ [mm] in opening.

Its weight is 46[kg].



## EPS

The amount of electric power supply is 78[W] on 28[V].

# Specification of ORBIS

## ACS

A method of attitude control is zero-momentum in 3 axis.

## COMM

ORBIS adopt Amateur band and S-band.

Amount of communication are

84[MB/day] in Amateur Band and

21[MB/day] in S-band.

## Orbit

LEO on 550[km] / Inclination is 31[deg] / Orbital period is 96[min]

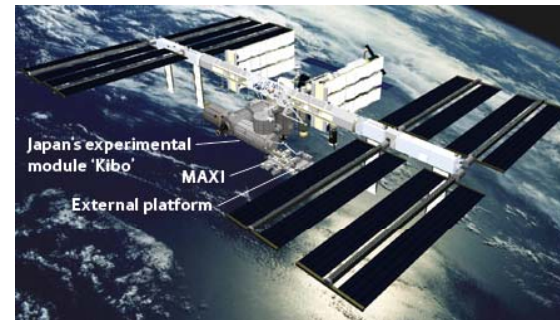
# Microsatellites for Science Mission

Science missions using **existing satellites** or **ground equipment**

- ✓ cannot take **enough time for observation** and
- ✓ if they can, their **sensitivity will be lower.**



Proposal for ground equipment  
is very hard



Sensitivity of MAXI is not enough  
for certain science mission

**Microsatellites are best suited** for science missions  
which needs both **long term** and **high-sensitivity** observation.