Development of Binary Black Hole Observation Satellite “ORBIS”

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Agenda

• Concept of ORBIS
• Mission of ORBIS

Progress of
• Mission Subsystem
• Attitude Determination Subsystem
• Structure Subsystem
• other Subsystem

• Summary
Microsatellite for Science Mission

Space Systems Laboratory in Tokyo Metropolitan University is developing a general-purpose microsatellite with ease of integration for science missions, because

✓ microsatellites are the most promising infrastructure for certain science missions.
✓ and the specification for general-purpose application will make satellites inexpensive and trusted.
We are developing ORbiting Binary black hole Investigation Satellite, ORBIS, as the first one of the general-purpose microsatellites for science mission.
Microsatellite for Science Mission

Microsatellites are the most promising infrastructure for certain science missions.

because

✓ there is no way in the current astronomy to accomplish such science missions as conduct long-term observations for the specified astral bodies with respect to budget, period, and occupancy time.
✓ and some scientists expect an utilization of microsatellite for the observations.

Investigation of Binary Black Hole is one of these science mission which is eagerly requiring microsatellite.
Mission of ORBIS

Binary Black Hole (BBH) consists of a pair of black holes and would contribute to develop kinds of cosmology. Past researches are saying that BBH surely exist but has not been observed yet.

BBH is dark object, and shows unique cycle variation in X-ray region.

which requires
✓ long term, continuous
✓ and high-precision observation.

So that microsatellite is and only a way for BBH observation.
Architecture for ORBIS is based on distributed autonomous system, where each subsystem is modularized into a block and communicates with the others via CAN bus.

Though this architecture is partially realized by Panel ExTenstion SATellite (PETSAT), developed by SOHLA, ORBIS adds another subsystem, Bulletin Board subsystem, to make reduction in labor required for system integration and actively utilize Flag & Play.
ORBIS Project

Concept of ORBIS won the first prize at 18th Satellite Design Contest in 2010, and we officially started ORBIS Project.

The project team consists of members in TMU, Kogakuin University, Meisei University and ISAS/JAXA.

Now, ORBIS is on BBM and Structure Thermal Model phase.
Progress of Mission Subsystem

Mission Subsystem with X-ray observation device is designed to be installed into the mission box in ORBIS. The incident light is concentrated with a kind of concentration lens to be focused on the CCD.
✓ We conducted some tests on Peltier device in a vacuum chamber, and the result CCD can be cooled to -26[deg]
✓ and now we are designing a light-collecting system and circuits for light-receiving system.
Progress of ADS

For highly-precise observation, Attitude Determination Subsystem adopts Star Tracker and Gyroscopes. In order to judge whether certain gyroscope can be use for ORBIS or not, we conducted a static experiment on a gyroscope.

As a result, **Estimate Accuracy of Attitude Angle** is $9.3 \times 10^{-4}$ [deg] on the gyroscope, showing the gyroscope can satisfy ORBIS’s requirement, less than $1.0 \times 10^{-3}$, at native environment.
We designed Structure Thermal Model of ORBIS and conduct series of vibration tests on it under the vibration levels of Qualification Test of H-IIA. The result gave us important information to improve the structure.
Progress of other subsystem

✓ C&DH conducted a communication test on CAN bus and now is designing common circuit board of ORBIS which is installed into all the subsystems.

✓ EPS estimated the amount of power supply and consumption to distribute power to all the subsystems.

✓ ACS analyzed disturbance of reaction wheels and now developing the simulator of entire ADCS.
Progress of other subsystem

✓ COMM analyzed the data traffic for Amateur band and S-band and tested some of them.

✓ TCS estimated the amount of thermal input, where the evaluation of contact thermal resistance in the most important and will be tested with the model.
Summary

- Space Systems Laboratory in Tokyo Metropolitan University is developing a general-purpose microsatellite for science mission, ORBIS.

- We set investigation of the Binary Black Hole as its mission.

- ORBIS is now under BBM phase and each subsystem is developing and testing their own module.
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Specification of ORBIS

Structure
ORBIS has 4 paddles of EPS.
Its dimension is $460 \times 460 \times 440$[mm] in closing and $980 \times 980 \times 440$[mm] in opening.
Its weight is 46[kg].

EPS
The amount of electric power supply is 78[W] on 28[V].
Specification of ORBIS

ACS
   A method of attitude control is zero-momentum in 3 axis.

COMM
   ORBIS adopt Amateur band and S-band.
   Amount of communication are
   84[MB/day] in Amateur Band and
   21[MB/day] in S-band.

Orbit
   LEO on 550[km] / Inclination is 31[deg] / Orbital period is 96[min]
Microsatellites for Science Mission

Science missions using existing satellites or ground equipment
✓ cannot take enough time for observation and
✓ if they can, their sensitivity will be lower.

Microsatellites are best suited for science missions which needs both long term and high-sensitivity observation.